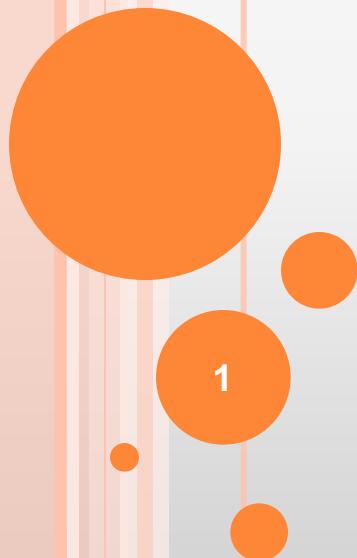




Gas Turbine Cycle

Lecture 2 Simple gas cycle

Tikrit university\ engineering college\ mechanical dept.





Air standard analysis

Using the perfect gas relations, we can write equations (1 & 3) in terms of pressure ratio (r_p) across the turbine:

$$\text{- pressure ratio } (r_p) = \frac{p_3}{p_4} = \frac{p_2}{p_1}$$

Using the isentropic relations :-

The absolute temp ratio across the compressor:

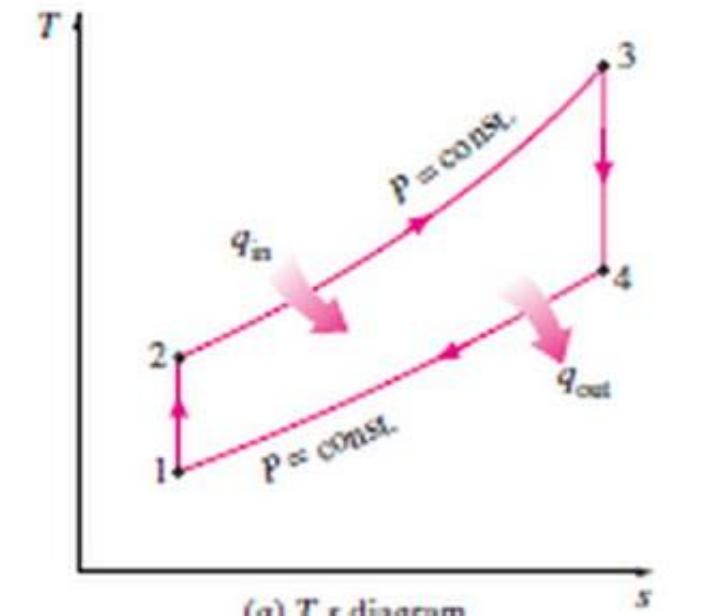
$$\frac{T_2}{T_1} = \left(\frac{P_2}{P_1}\right)^{\frac{k-1}{k}} = (r_p)^{\frac{k-1}{k}} \dots \dots \dots (4.a)$$

Where: $k = \frac{c_p}{c_n} = 1.4$ (for cold air)

And across the turbine is given as :-

$$\frac{T_3}{T_4} = \frac{P_3}{P_4}^{\frac{k-1}{k}} = (r_p)^{\frac{k-1}{k}} \dots \dots \dots \quad (4.b)$$

$k = 1.33$ (for hot gas)





So that the specific work of the turbine from eq.3 is given as :-

$$W_T = C_p T_3 \left(1 - \frac{T_4}{T_3}\right) = C_p T_3 \left(1 - \frac{1}{(r_p)^{\frac{k-1}{k}}}\right) \quad \dots \dots \dots \quad (5)$$

Where $T_3 \neq T_2$

And the specific work of compressor from eq.1 is given as :-

$$W_c = C_p T_2 \left(1 - \frac{T_1}{T_2} \right) = C_p T_2 \left(1 - \frac{1}{(r_p)^{\frac{k-1}{k}}} \right) \dots \dots \dots \quad (6)$$

where $(r_p)_c = (r_p)_T = \frac{p_2}{p_1} = \frac{p_3}{p_4} = \text{constant}$.

The net specific work of the gas turbine cycle :-

And the output power, $P = \dot{m} * W_{\text{net}}$ (8)



The thermal efficiency of the cycle is:

$$\eta_{th} = \frac{W_{net}}{Q_{add}} = \frac{W_T - W_C}{Q_{add}} \quad \dots \dots \dots \quad (9)$$

$$= \frac{C_p(T_3 - T_4) - C_p(T_2 - T_1)}{C_p(T_3 - T_2)} = \frac{C_p(T_3 - T_2) - C_p(T_4 - T_1)}{C_p(T_3 - T_2)}$$

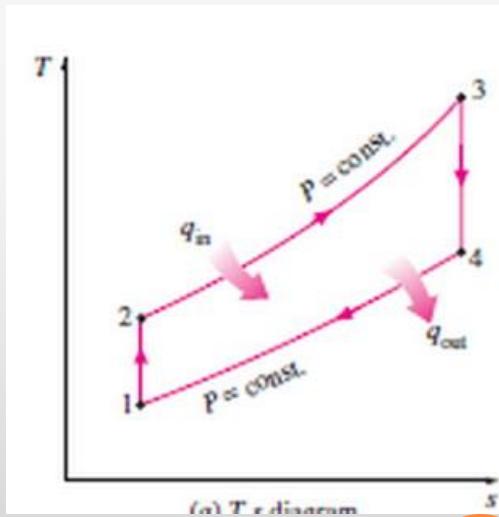
$$= 1 - \frac{(T_4 - T_1)}{(T_3 - T_2)} = 1 - \frac{T_1}{T_2} \left[\frac{\left(\frac{T_4}{T_1} \right) - 1}{\left(\frac{T_3}{T_2} \right) - 1} \right], \text{ where } \frac{T_4}{T_1} = \frac{T_3}{T_2}$$

Cold air standard (constant specific heat).

$$W_c = C_p (T_2 - T_1)$$

Air – standard (variable specific heat).

$$W_c = h_2 - h_1$$



Example : Gas turbine unit has pressure ratio ($r_p = \frac{6}{1}$) & max temp is 600°C. The isentropic efficiency ($\eta_c = 0.82$, $\eta_T = 0.85$). Calculate the power output in (kW), the flow rate of the hot gasses entering the turbine at rate of $15 \frac{kg}{s}$. The air entrance temp to the compressor is 15 °C, and find BWR.

$$\text{for air : } C_p = 1.005 \frac{kJ}{kg.K}, k = 1.4 .$$

$$\text{For gas : } C_p = 1.13 \frac{kJ}{kg.K}, k = 1.33 .$$

Solution:

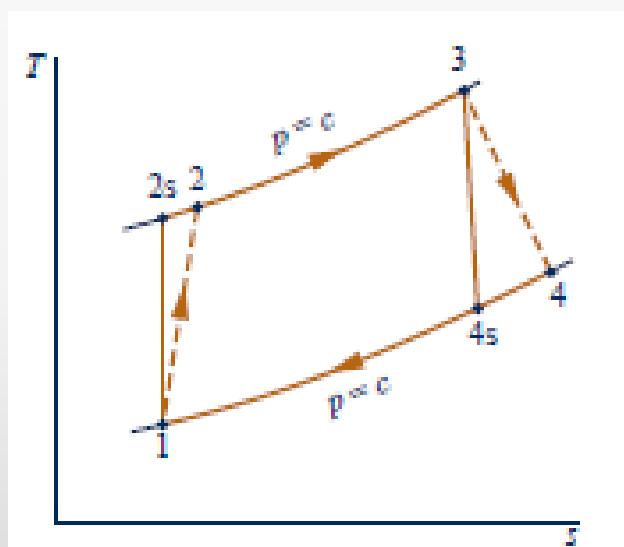
For isentropic compression process (compressor):

$$\frac{T_{2s}}{T_1} = \left(\frac{P_2}{P_1}\right)^{\frac{k-1}{k}}$$

$$T_1 = 15 + 273 = 288 \text{ K}$$

$$\therefore T_{2s} = 288 * (6)^{\frac{1.4-1}{1.4}} = 481 \text{ K}$$

$$\eta_c = \frac{\text{ideal work}}{\text{actual work}} = \frac{T_{2s}-T_1}{T_2-T_1} = 0.82$$



$$T_2 = 521 \text{ K}$$

- for isentropic expansion process (turbine):

$$\frac{T_3}{T_{4s}} = \left(\frac{P_3}{P_4}\right)^{\frac{k-1}{k}} = (6)^{\frac{1.33-1}{1.33}}, \text{ where } T_3 = 600 + 273 = 873 \text{ K}$$

$$T_{4s} = 558 \text{ K}$$

$$\eta_T = \frac{\text{actual work}}{\text{ideal work}} = \frac{T_3 - T_4}{T_3 - T_{4s}} = 0.85$$

$$T_4 = 605 \text{ K}$$

$$w_C = C_p(T_2 - T_1) = 1.005(521 - 288) = 234.16 \text{ kJ/kg}$$

$$w_T = C_p(T_3 - T_4) = 1.11(873 - 605) = 297.48 \text{ kJ/kg}$$

$$W_{\text{net}} = w_T - w_C = 63.31 \text{ kJ/kg}$$

$$\text{Power output} = \dot{m} * w_{\text{net}} = 15 * 63.31 = 949.72 \text{ kW}$$

$$\eta_{\text{th}} = \frac{w_{\text{net}}}{q_{\text{add}}} = \frac{63.31}{C_p(T_3 - T_2)} = \frac{63.31}{1.11(873 - 521)} = 15.8 \%$$

$$\text{BWR} = \frac{w_C}{w_T} = \frac{C_p(T_2 - T_1)}{C_p(T_3 - T_4)} = 0.787$$

